

GMES Sentinel-2

The Optical High Resolution Mission for GMES Operational Services

16 July 2007, FAO, Roma

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Sentinel-2 Presentation, FAO, 16 July 2007

Global Monitoring for the Environment and Security

**European independence in critical data sources
for environmental monitoring and security**

and

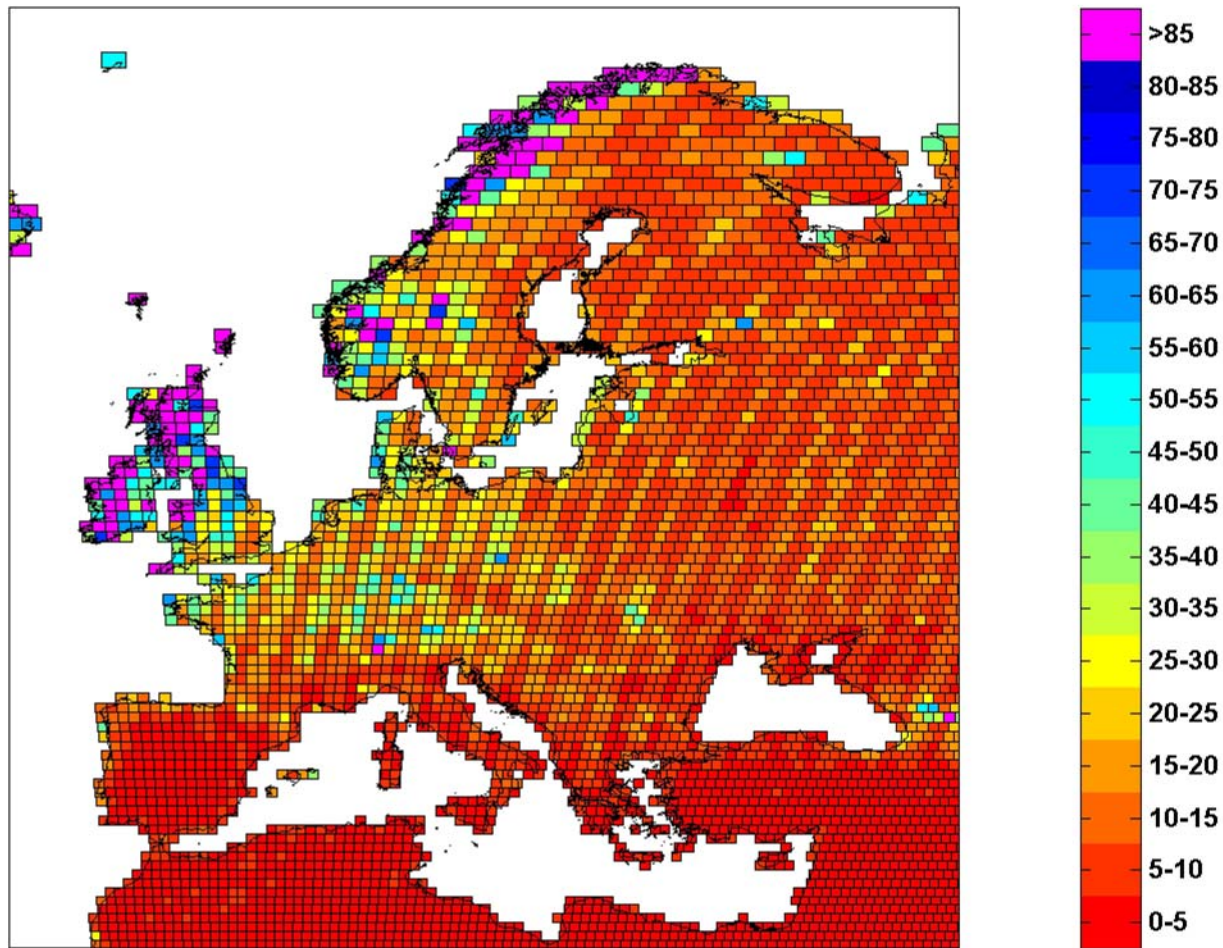
**the European contribution to the Global Earth
Observation System of Systems (GEOSS)**

Sentinel-1, Sentinel-2, Sentinel-3

- **Full and systematic coverage of land surface (including major islands) globally from -56° (Southern America) to $+83^{\circ}$ (Northern Greenland) latitude. To achieve this objective and to provide high mission availability, a constellation of 2 operational satellites is required, allowing to reach a 5 day geometric revisit time. The revisit time with only one operational satellite as it will be the case at the beginning of the deployment of the system is 10 days.**
- **Additionally, the capability to access more rapidly (within 1 to 3 days) some limited geographical areas in emergency mode by performing a roll-tilt manoeuvre of the spacecraft has been included in the design to fulfil emergency services and will be used to observe major disaster (e.g. floods, earthquakes) at their occurrence.**

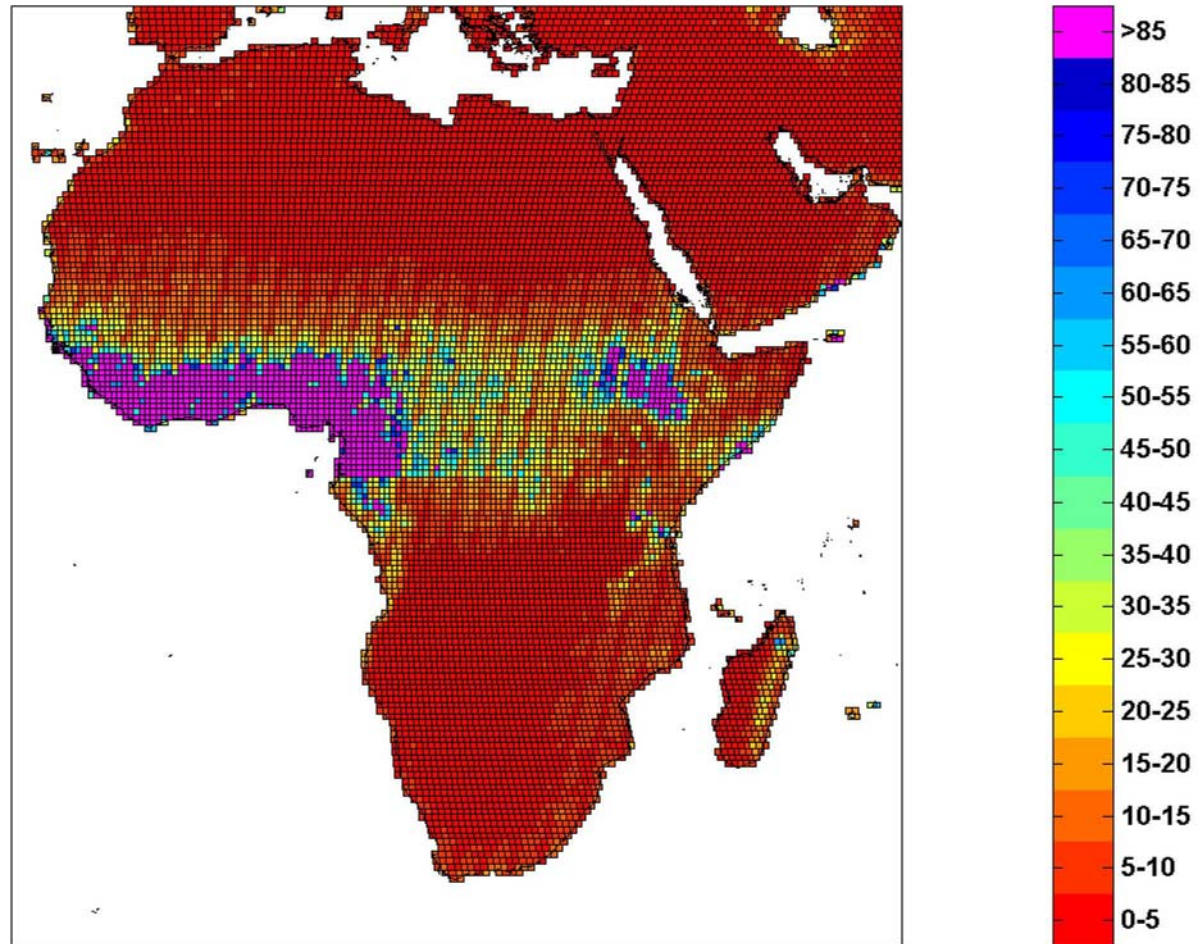
Revisit time over Europe in summer with 2 satellites

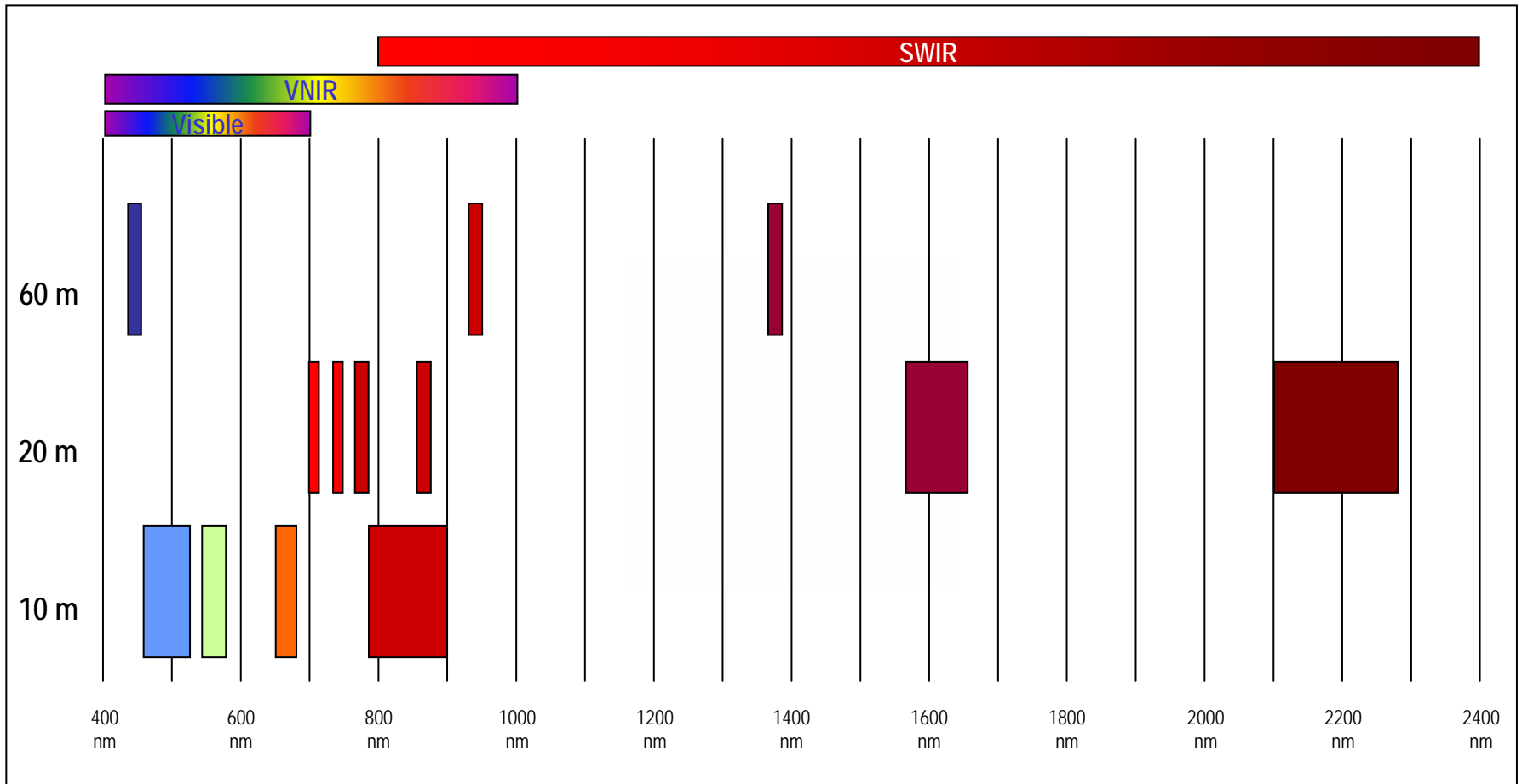
Maximum effective coverage time for SC1 & SC2 (days) (<15% cloud cover; 68% confidence)



Revisit time over Africa in summer with 2 satellites

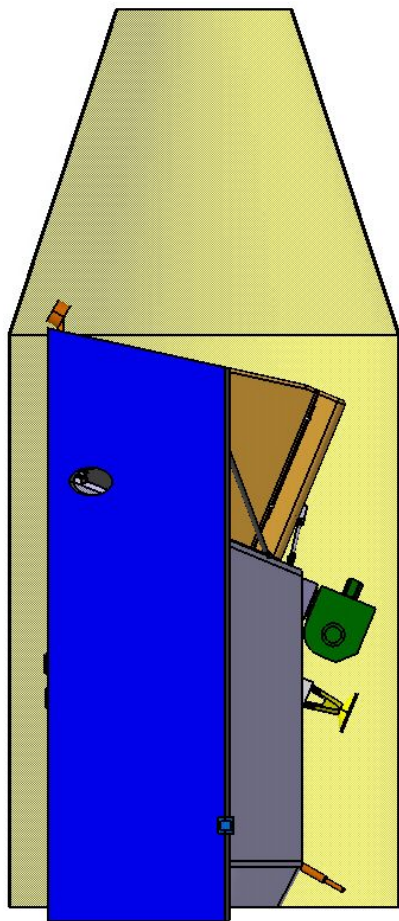
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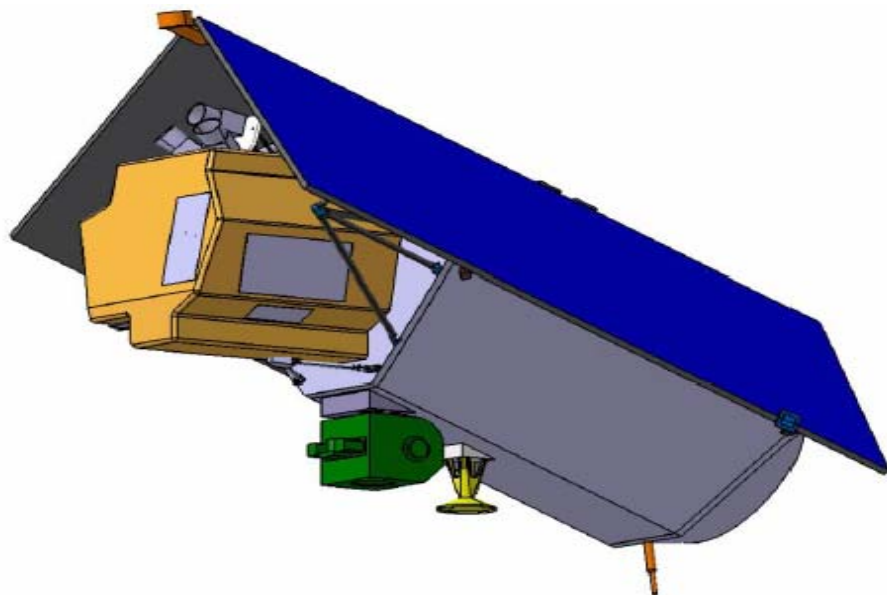


13 spectral bands versus spatial sampling distance

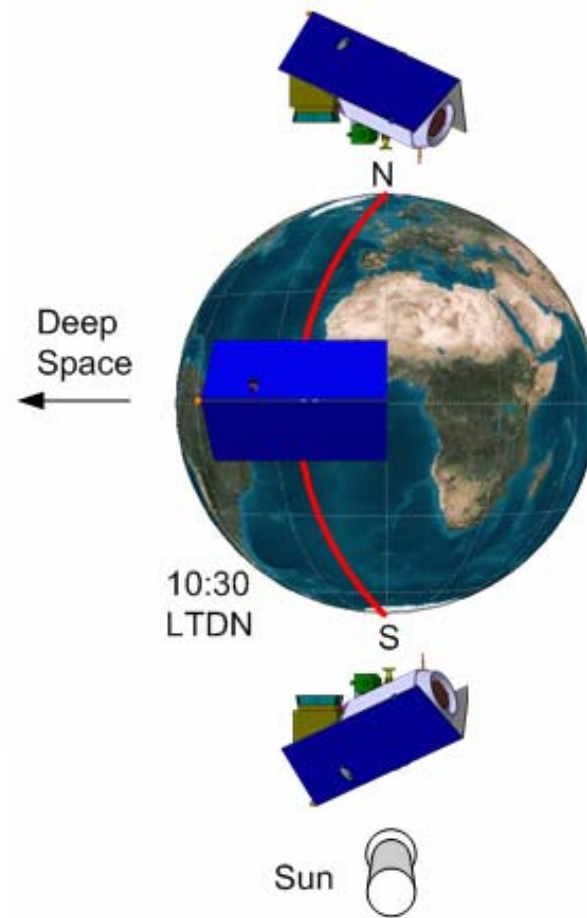
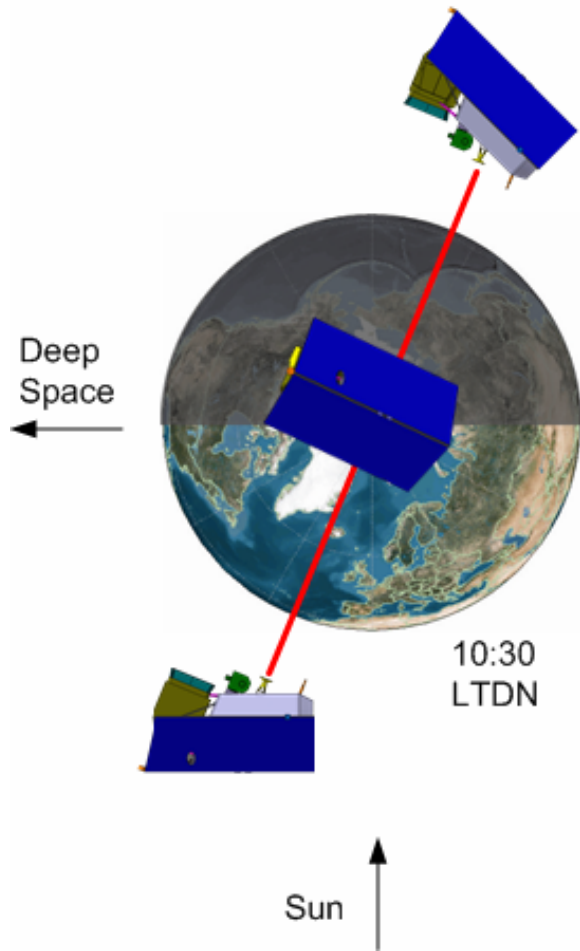
- **Compact satellite with about 1 ton mass.**
- **7 year design lifetime, with propellant for 12 year operations.**
- **Roof-shaped configuration with a fixed body-mounted solar array, as considered in phase A/B1, leads to a simple mechanical design.**
- **3-axis stabilized with AOCS based on high-rate multi-head star trackers, gyrometer and GNSS receiver.**
- **The power subsystem relies on high efficiency solar cells (GaAs triple junction) and a Li-ion battery.**
- **2 Terabit solid state mass memory, payload data downlink at a rate of 450 Mb/s in X-band with a high spectral efficiency modulation (8PSK).**
- **Command and control performed with omni-directional S-band antenna coverage.**



Satellite within VEGA fairing

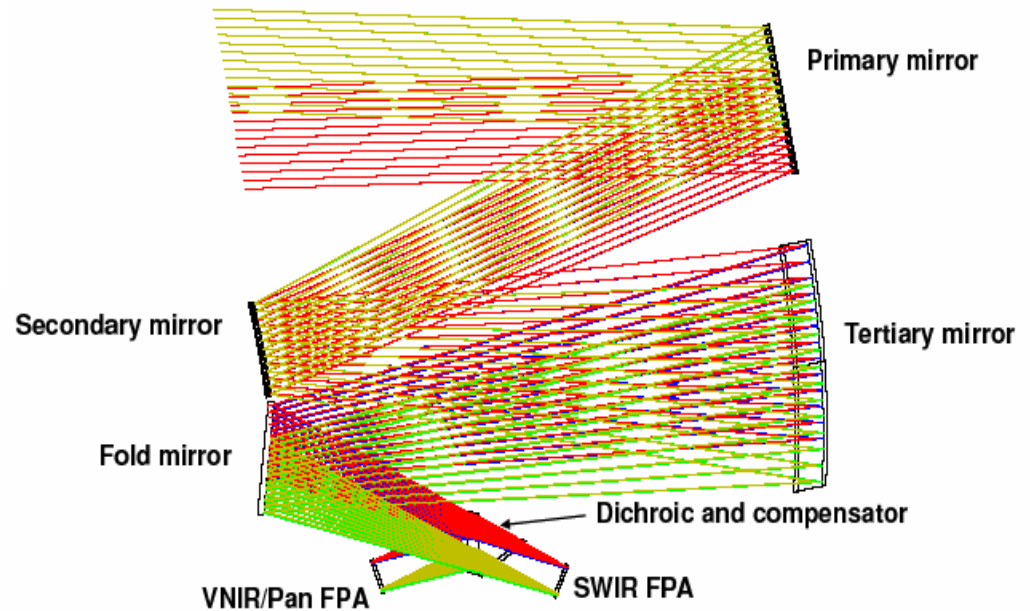
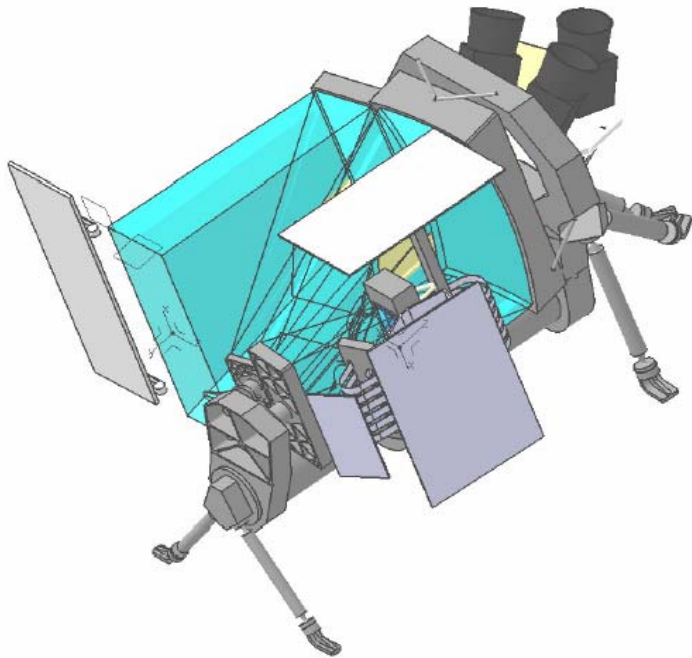


Satellite in-orbit configuration



In its orbital configuration, the satellite is tilted by 22.5° with respect to the roll axis in order to maximize the solar energy collected by the solar array.

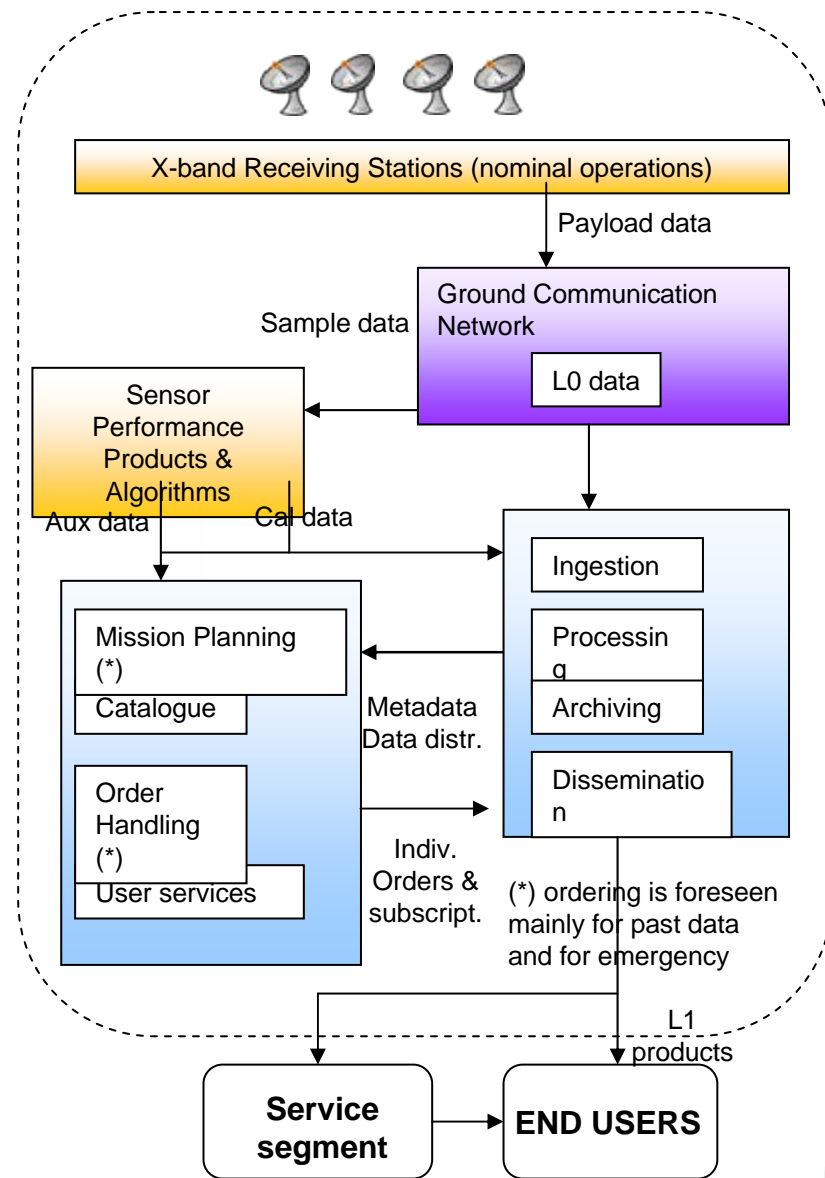
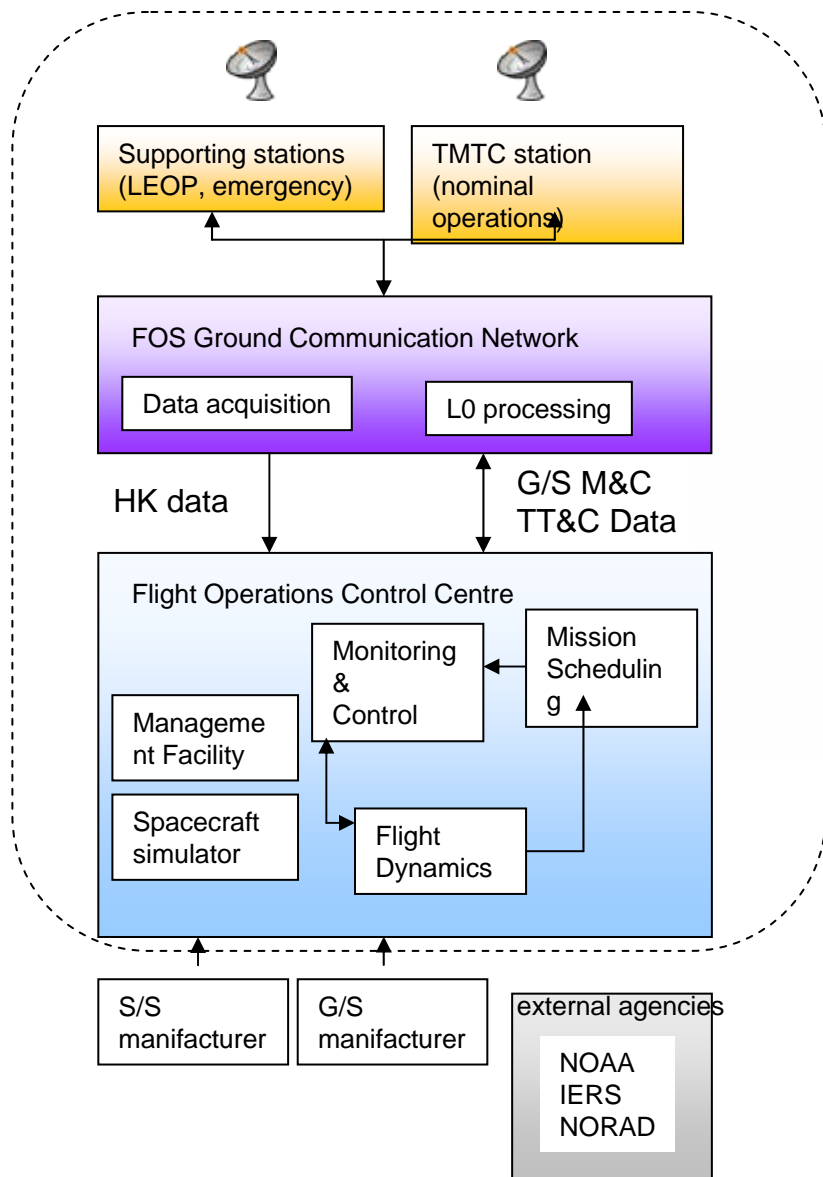
- Push-broom instrument, 200 kg mass, 170 W power consumption
- Three Mirror Anastigmat (TMA) telescope with **150 mm pupil diameter**, providing a very good imaging quality all across the wide field of view, corresponding to **290 km swath width**, significantly enlarged with respect to Landsat (185 km) and SPOT (120 km). Telescope structure and mirrors are made of silicon carbide to minimize thermo-elastic deformations.
- VNIR focal plane based on monolithic **CMOS detectors**, SWIR Focal plane based on a mercury cadmium telluride (MCT) detector hybridised on a CMOS read-out circuit. A dichroic beam-splitter provides the spectral separation of VNIR and SWIR channels.
- A combination of partial on-board calibration, using a **sun diffuser**, and vicarious calibration with ground targets will guarantee a high quality radiometric performance. The detector signals are converted to digital with **12-bit resolution** and state-of-the-art data compression, based on wavelet transform, is applied to reduce the data volume.
- A shutter mechanism is implemented to prevent the instrument from direct viewing the sun in orbit and from contamination during launch.
- The average observation time per orbit is about 16 minutes, while the peak value is 31 minutes.

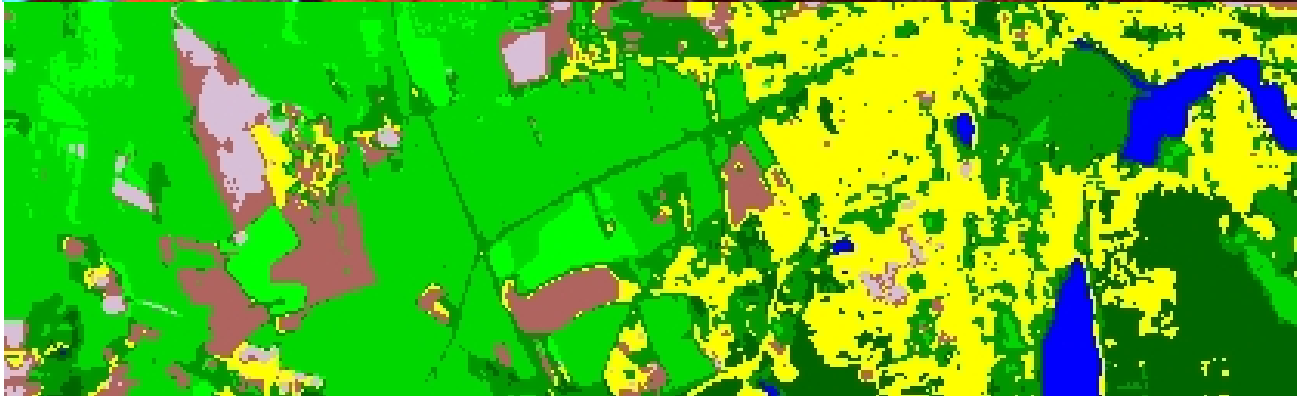
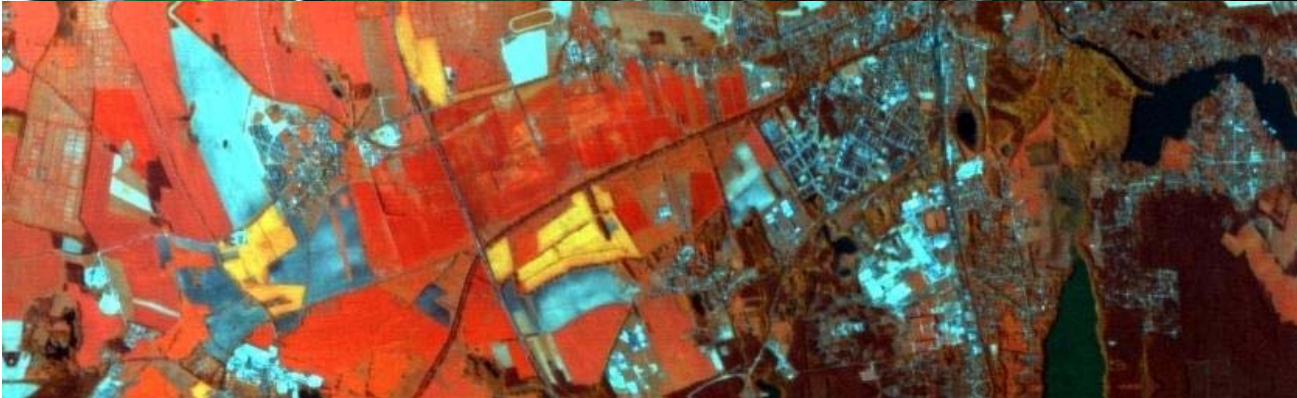


Multi-Spectral Instrument View









MSI optical design across view

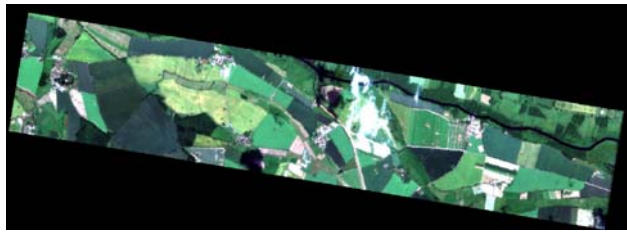
In order to mitigate the development risks and secure the development schedule, technology pre-development activities have been initiated on critical items, namely the VNIR and SWIR detectors and the filter assemblies.



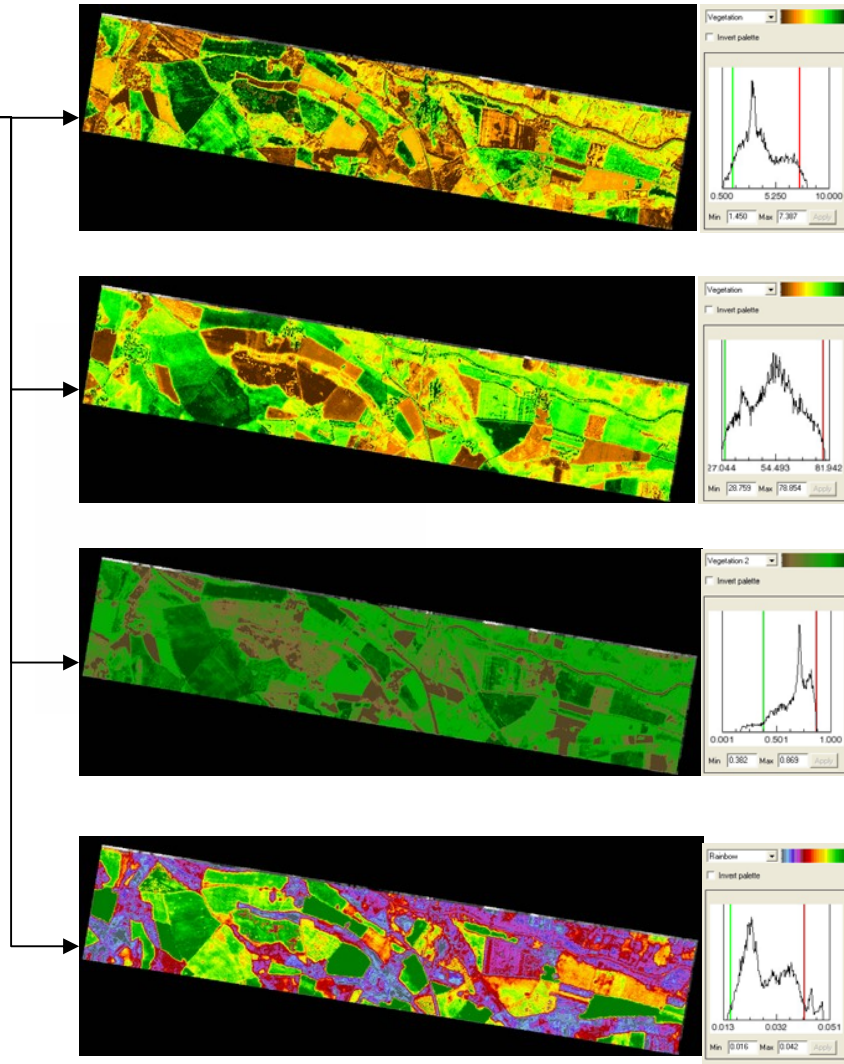


Simulated Sentinel-2 colour composite images using the red/green/blue bands (above) and near-infrared/red/green (below)

	Urban
	Water
	Forest 1
	Forest 2
	Bare soil 1
	Bare soil 2
	Cultivated field 1
	Cultivated field 2



Sentinel-2 simulated Level 1

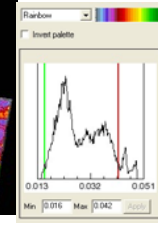
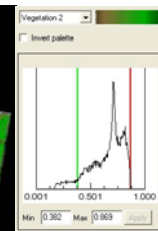
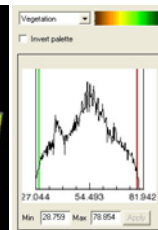
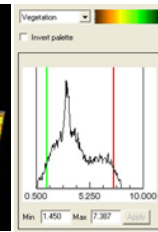


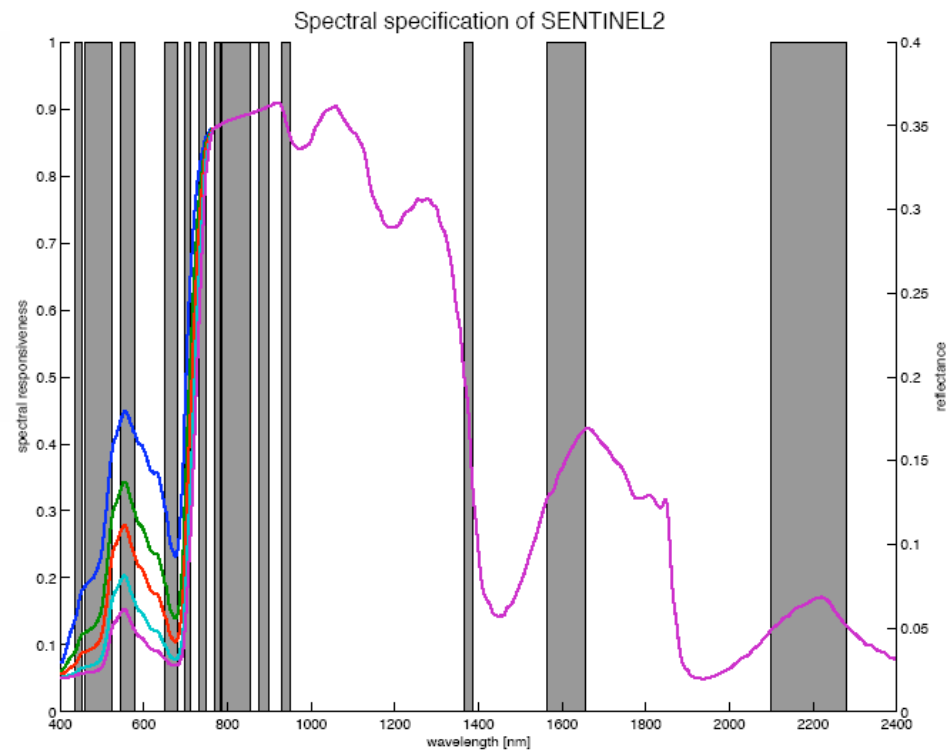
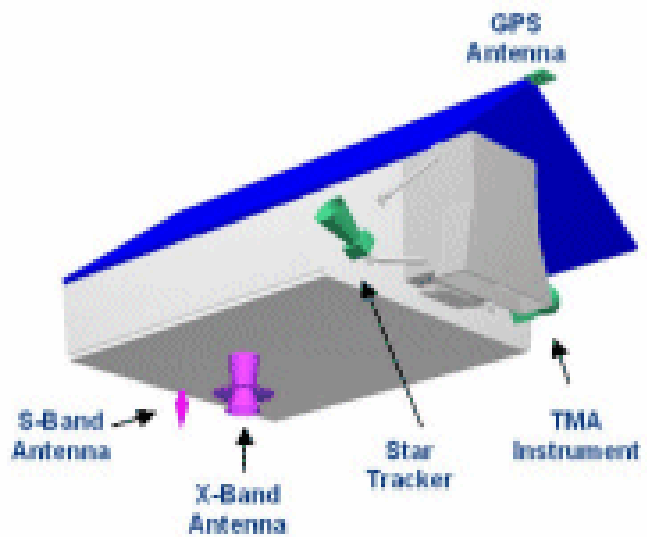
Leaf Area Index

Leaf chlorophyll content

Fractional cover

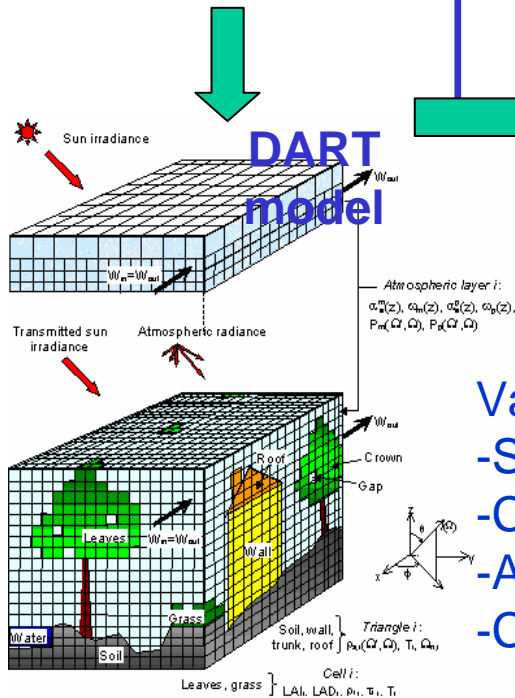
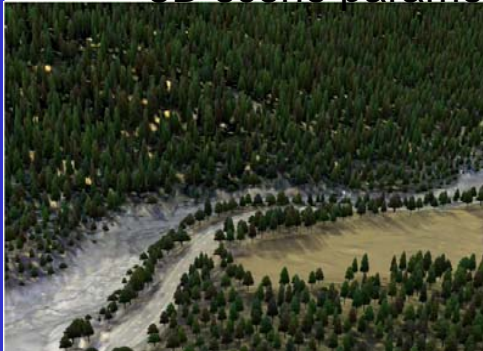
Leaf water content





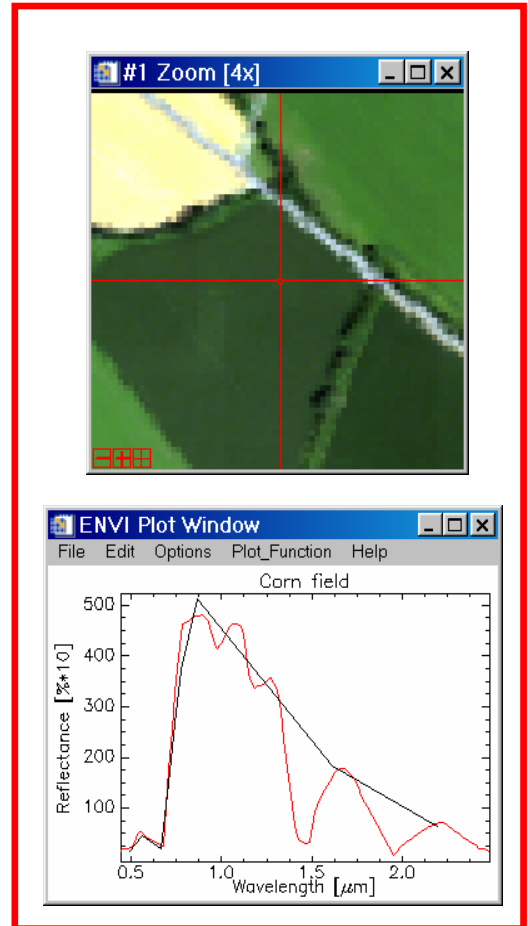
- Evaluation of SENTINEL 2 specifications relevant for level 2 products
 - spectral, geometric, temporal
- Simulation of synthetic scenarios
 - ⇒ 3D scene reconstruction with the DART model
 - ⇒ Changing environmental & atmospheric conditions
 - ⇒ Relevant for selected user services
 - ⇒ Validation with real imaging spectrometer data

3D scene parameterization



Variable Specifications:

- Sensor specs
- Calib. & instr. Error
- Atmosphere & illumination
- Compression factor



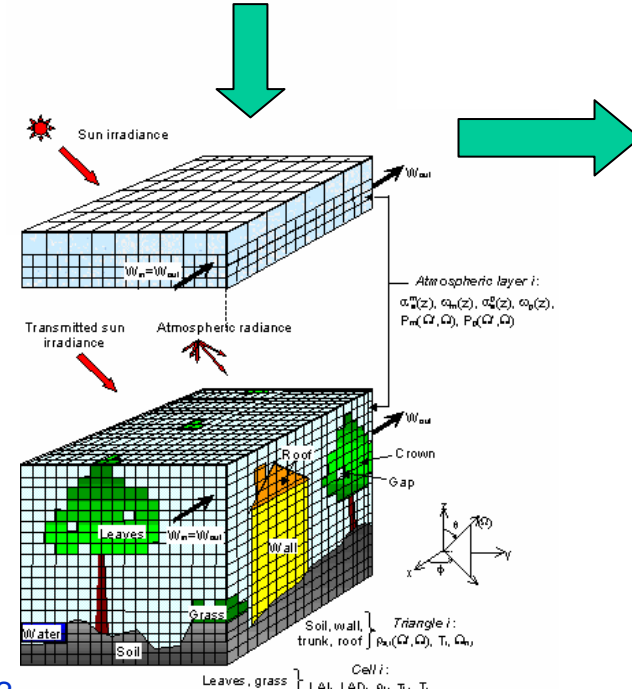
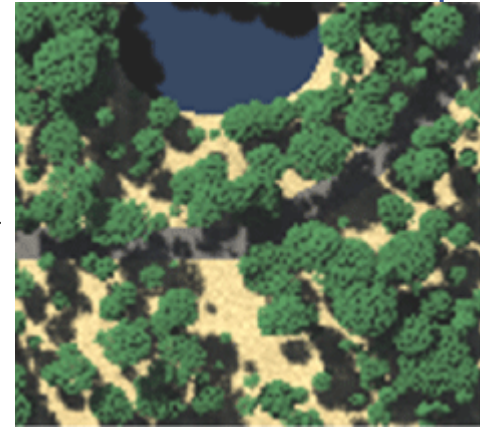
3D landscape:
 -LIDAR
 -Imaging spectrometry
 Optical properties:
 -Field spectrometry
 Canopy structure
 -Field measurements

DART model:
 -Radiative Transfer Model
 -Spatial scene representation
 -Natural / urban landscapes

3D scene parameterization



Example: Forest scene



Synthetic scene:
 -Top-of-canopy reflectance
 -Diff. view angles
 -variable atmosphere & illumination

- The Sentinel-2 wide-swath high-resolution multispectral system will provide enhanced continuity to the SPOT- and Landsat-type observations - with improved revisit time, coverage area, spectral bands, swath width, radiometric and geometric image quality.
- In terms of programmatics, ESA has performed the Sentinel-2 Definition Phase A/B1 in 2005 and 2006 with an industrial consortium led by Astrium GmbH. Following the successful completion of this Phase A/B1 in January 2007, the Invitation to Tender for the Implementation Phase was released in February 2007 and proposal received in May 2007.
- The Implementation Phase B2/C/D/E1 is expected to start in **October 2007** and the **launch of the first satellite is foreseen for 2012.**